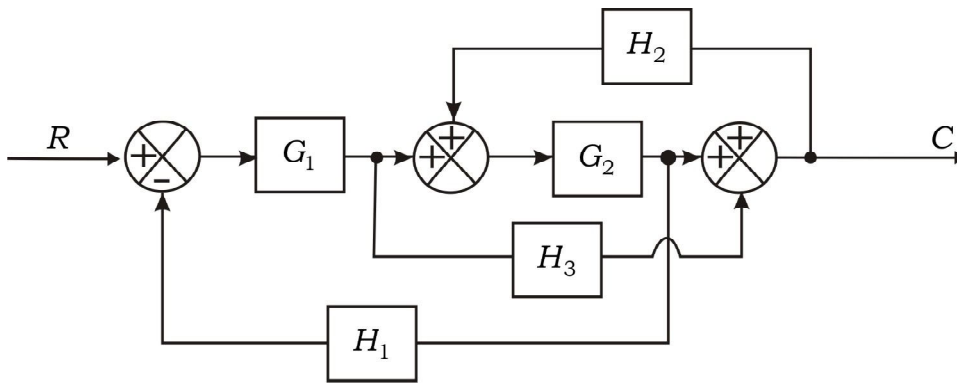


## Test Series: Control System with Solution

**Q.1** For the system shown the equivalent T.F. is



(a)  $G_1 \left[ \frac{G_2 + H_3}{1 + G_1 G_2 H_1 - G_2 H_2} \right]$

(b)  $\frac{G_1 G_2 + G_2 H_3}{1 + G_2 H_1 - G_1 G_2 H_2}$

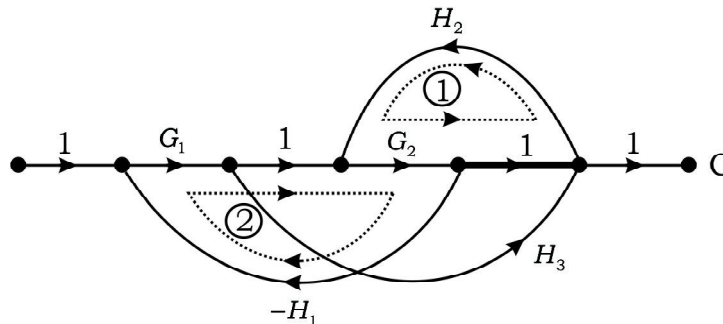
(c)  $\frac{G_2 H_3}{1 + G_2 H_1 + G_1 G_2 H_2}$

(d)  $\frac{G_1 G_2 H_3}{1 - G_1 H_1 - G_2 H_2 H_3}$

**Ans. (a)**

**Exp.**

The signal flow graph is



Number of forward paths  $n = 2$

Gain along 1<sup>st</sup> forward path

$$M_1 = 1 \cdot G_1 \cdot 1 \cdot G_2 \cdot 1 \cdot 1 = G_1 G_2$$

$$M_2 = 1 \cdot G_1 \cdot H_3 \cdot 1 = G_1 H_3$$

Two touching loops

$$\Delta = 1 - \{-G_1 G_2 H_1 + G_2 H_2\}$$

$$\Delta_1 = 1, \Delta_2 = 1$$

$$T.F. = \frac{M_1 \Delta_1 + M_2 \Delta_2}{\Delta} = \frac{G_1 G_2 + G_1 H_3}{1 + G_1 G_2 H_1 - G_2 H_2}$$

**Q.2** Consider a ufb system with forward path T.F. is  $G(s) = \frac{K(s+2)(s+4)}{(s-1)(s-3)}$ . The range of  $K$  for which system is unstable

## Test Series: Control System with Solution

(a)  $K < \frac{2}{3}$

(b)  $K > -1$

(c)  $-1 < K < \frac{2}{3}$

(d)  $-1 > K$  or  $K > \frac{2}{3}$

**Ans. (c)**

**Exp.**

The characteristic equation  $1 + G(S) = 0$

$$(s - 1)(s - 3) + K(s + 2)(s + 4) = 0$$

$$(K + 1)s^2 + (6K - 4)s + (8K + 3) = 0$$

Routh Hurwitz table

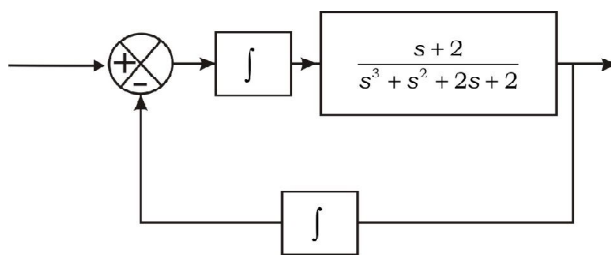
$$\begin{array}{c|cc} s^2 & K + 1 & 8K + 3 \\ s^1 & 6K - 4 & 0 \\ s^0 & 8K + 3 & 0 \end{array}$$

For stable system there should not be any sign change in 1<sup>st</sup> column either all -ve or all +ve

$$\left. \begin{array}{l} K + 1 < 0 \Rightarrow K < -1 \\ 6K - 4 < 0 \Rightarrow K < 2/3 \\ 8K + 3 < 0 \Rightarrow K < -3/8 \end{array} \right\} \Rightarrow K < -1$$

$$\left. \begin{array}{l} K + 1 > 0 \Rightarrow K > -1 \\ 6K - 4 > 0 \Rightarrow K > 2/3 \\ 8K + 3 > 0 \Rightarrow K > -3/8 \end{array} \right\} \Rightarrow K > 2/3$$

**Q.3** For the system shown



The number of poles in RH & on  $j\omega$ -axis are

(a) 0 & 0

(b) 2 & 0

(c) 3 & 2

(d) 3 & 0

**Ans. (b)**

**Exp.**

Integrator  $\int = \frac{1}{s}$

$$G(s) = \frac{s + 2}{s(s^3 + s^2 + 2s + 2)}$$

## Test Series: Control System with Solution

$$H(s) = \frac{1}{s}$$

Ch. equation  $1 + G(s)H(s) = 0$

$$1 + \frac{s+2}{s^2(s^3+s^2+2s+2)} = 0$$

$$\Rightarrow s^5 + s^4 + 2s^3 + 2s^2 + s + 2 = 0$$

$s^5$	1	2	1
$s^4$	1	2	2
$s^3$	$0(\varepsilon)$	-1	0
$s^2$	$\frac{2\varepsilon+1}{\varepsilon}$	2	0
$s^1$	$-\frac{2\varepsilon^2+\varepsilon+1}{2\varepsilon+1}$	0	0
$s^0$	2	0	0

The 1<sup>st</sup> column if  $\varepsilon \rightarrow 0$  ( $\varepsilon \geq 0$ )

1	+ve
1	+ve
$\varepsilon$	+ve
$2 + \frac{1}{\varepsilon}$	+ve
$-\frac{2\varepsilon^2 + \varepsilon + 1}{2\varepsilon + 1}$	-ve
2	+ve

there are two sign changes so 2-roots are in R.H. of s-plane.

Complete - zero row is not observed so no root on  $j\omega$  - axis

**Q.4** The O.L.T.F for a *ufb* system is  $\frac{K}{s^2 + 2s - 3}$ . The value of K for which two closed

loop poles are lying on R.H. side of line  $s = -2$

(a)  $K > 3$

(b)  $K < 3$

(c)  $K > 0$

(d)  $K < 0$

**Ans. (a)**

**Exp.**

The Ch. equation

$$1 + \frac{K}{s^2 + 2s - 3} = 0$$

$$\Rightarrow s^2 + 2s + (K - 3) = 0$$

Let  $z = s + 2$

## Test Series: Control System with Solution

$$\Rightarrow s = z - 2$$

$$(z - 2)^2 + 2(z - 2) + K - 3 = 0$$

$$\Rightarrow z^2 - 2z + (K - 3) = 0$$

R.H. Table

$z^2$	1	$K - 3$
$z^1$	-2	0
$z^0$	$K - 3$	0

For two roots are lying on R.H. side of  $s = -2$  i.e. for which

$$\text{Re}[s] > -2 \Rightarrow \text{Re}[z] > 0$$

Hence two roots are lying on R.H.S. of z-plane  $z > 0$  so there should be two sign changes Hence

$$K - 3 > 0 \Rightarrow K > 3$$

**Q.5** The OLTF of a *ufb* system is  $G(s) = \frac{K}{s^m(s + \alpha)}$ .

The system has a setting time 0.4 sec and position error constant 10.

The value of  $K$  &  $\alpha$  are

(a) 10 & 10

(b) 100 & 100

(c) 10 & 100

(d) 100 & 10

**Ans. (d)**

**Exp.** As position error constant  $K_p = 10$  i.e. finite number Hence it is a type -0 system, i.e.  $m = 0$

$$G(s) = \frac{K}{s + \alpha} = \frac{K / \alpha}{1 + s / \alpha}$$

Compare with  $G(s) = \frac{K_p}{1 + sT}$

$$K_p = \frac{K}{\alpha} = 10, \quad T = \frac{1}{\alpha}$$

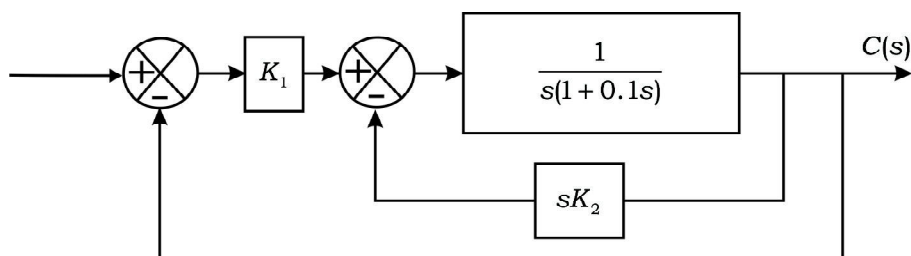
$$\text{Settling time } t_s = 4T = \frac{4}{\alpha} = 0.4$$

$$\Rightarrow \alpha = 10$$

$$\& K = 10\alpha = 100$$

**Common Date for Q. 6 and Q. 7**

For a control system shown



## Test Series: Control System with Solution

**Q.6** If  $K_2 = 0$ , determine the value of  $K_1$  for an steady state error of 0.1

- (a) 10 (b) 20  
(c) 5 (d) 6

**Ans. (a)**

**Exp.** If  $K_2 = 0$

$$OLTF = \frac{K_1}{s(1+0.1s)}$$

Compare with  $G(s) = \frac{K_v}{s(1+sT)}$

Velocity error constant  $K_v = K_1$

S.S error for ramp input

$$e_{ss} = \frac{1}{K_v} = \frac{1}{K_1} = 0.1$$

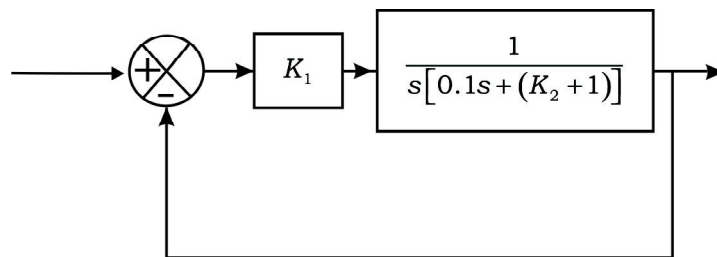
$\Rightarrow K_1 = 10$

**Q.7** If the damping ratio is to be changed to 0.8 without affecting S.S. error, the value of  $K_1$  &  $K_2$  are

- (a) 32, 2.2 (b) 25.6 & 1.56  
(c) 11.23, 1.12 (d) 10 & 0

**Ans. (b)**

**Exp.** Inner loop function =  $\frac{1}{1 + \frac{sK_2}{s(1+0.1s)}} = \frac{1}{s[K_2+1+0.1s]}$



$$OLTF = \frac{K_1}{s[(K_2+1)+0.1s]}$$

$$= \frac{K_1 / (K_2 + 1)}{s \left[ 1 + \frac{0.1}{K_2 + 1} s \right]}$$

Velocity error constant  $K_v = \frac{K_1}{K_2 + 1}$

## Test Series: Control System with Solution

$$e_{ss} = \frac{1}{K_v} = \frac{K_2 + 1}{K_1} = 0.1$$

$$\Rightarrow K_2 + 1 = 0.1 K_1$$

Characteristic eq<sup>n</sup>

$$1 + \frac{K_1}{s[(K_2 + 1) + 0.1s]} = 0$$

$$\Rightarrow 0.1s^2 + s(K_2 + 1) + K_1 = 0$$

$$s^2 + 10(K_2 + 1)s + 10K_1 = 0$$

Compare with  $s^2 + 2\rho\omega_n s + \omega_n^2 = 0$

$$\omega_n = \sqrt{10K_1}, \quad 2\rho\omega_n = 10(K_2 + 1)$$

$$\Rightarrow \rho = \frac{5(K_2 + 1)}{\sqrt{10K_1}} = 0.8$$

$$K_2 + 1 = 0.1K_1$$

$$\frac{5 \times 0.1 K_1}{\sqrt{10K_1}} = 0.8$$

$$\Rightarrow K_1 = 25.6$$

$$K_2 = 0.1K_1 - 1 = 1.56$$

**Q.8** The unit impulse open-loop system response of a system is  $g(t) = 1 - e^{-t}$ ,  $t \geq 0$ . Now if unity feedback is provided, the steady state errors for unit ramp and unit parabolic input are

(a) 1, 0

(b)  $\infty, \infty$

(c)  $\infty, 1$

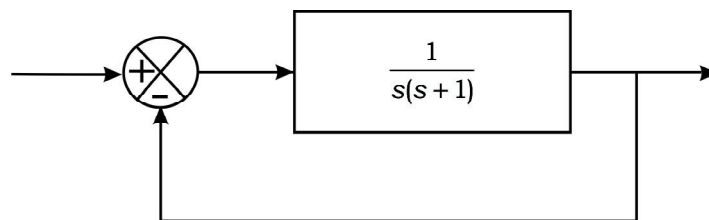
(d) 1,  $\infty$

**Ans. (d)**

**Exp.**  $g(t) = 1 - e^{-t}$

$$O.L.T.F. \quad G(s) = \mathcal{L}[g(t)] = \frac{1}{s} - \frac{1}{s+1}$$

$$= \frac{1}{s[1+s]}$$



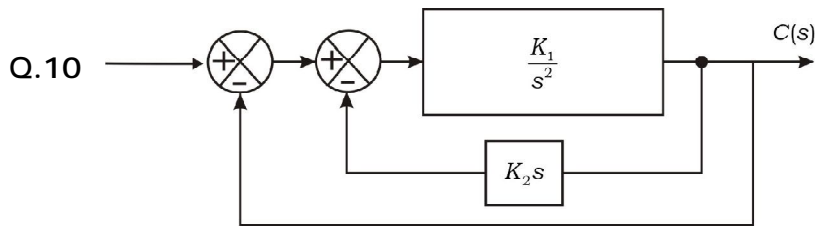
It is type -1 system with



## Test Series: Control System with Solution

$$\Rightarrow 0.99 = 1 - e^{-t_x}$$

$$\Rightarrow t_x = 4.6 \text{ sec.}$$



If the damping factor and settling time of the system are 0.8 & 5 rad/sec respectively. The values of  $K_1$  &  $K_2$  are respectively

(a) 5, 0.16

(b) 25, 0.32

(c) 10, 2

(d) 50, 2.5

**Ans. (b)**

**Exp.**

Inner loop T.F.  $G(s) = \frac{K_1/s^2}{1 + K_2s \cdot \frac{K_2}{s^2}}$

$$= \frac{K_1}{s^2 + K_1K_2s}$$

$$G(s) = \frac{K_1}{s(s + K_1K_2)}$$

Char. eq<sup>n</sup>  $1 + G(s) = 0$

$$1 + \frac{K_1}{s(s + K_1K_2)} = 0$$

$$\Rightarrow s^2 + K_1K_2s + K_1 = 0$$

Compare with  $s^2 + 2\rho\omega_n s + \omega_n^2 = 0$

$$\omega_n^2 = K_1, \quad 2\rho\omega_n = K_1K_2$$

$$\Rightarrow \omega_n = \sqrt{K_1}, \quad \rho = \frac{K_1K_2}{2\omega_n} = \frac{K_1K_2}{2\sqrt{K_1}}$$

$$\rho = \frac{K_2}{2}\sqrt{K_1}$$

$$\omega_n = \sqrt{14}$$

$$\Rightarrow K_1 = \omega_n^2 = 25$$

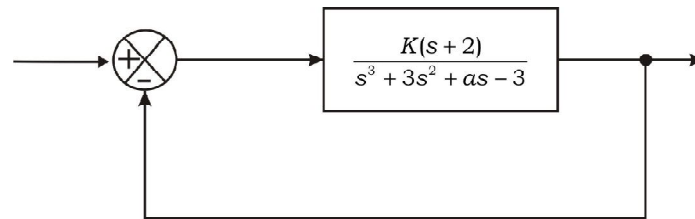
$$\rho = \frac{K_2}{2}\sqrt{K_1}$$

$$0.8 = \frac{K_2}{2} \times 5$$

## Test Series: Control System with Solution

$$\Rightarrow K_2 = 0.32$$

**Common Data for Q.11 & 12**



Where  $K > 0$  &  $a > 0$

**Q.11** The range of  $K$  for which the system is stable

(a)  $K < \frac{3}{2}$

(b)  $K > \frac{3}{2}$

(c)  $K > 0$

(d)  $K > 3$

**Ans. (b)**

**Q.12** The values of  $K$  &  $a$  so that the sustained oscillation occur

(a)  $3, \frac{5}{2}$

(b)  $\frac{3}{2}, 2$

(c)  $\frac{3}{2}, 5$

(d) Not possible

**Ans. (d)**

**Exp.** Ch. equation

$$1 + \frac{K(s+2)}{s^3 + 3s^2 + as - 3} = 0$$

$$\Rightarrow s^3 + 3s^2 + (K+a)s + (2K-3) = 0$$

$s^3$	1	$K+a$
$s^2$	3	$2K-3$
$s^1$	$\frac{K+3a+3}{3}$	0
$s^0$	$2K-3$	0

For stability

$$2K - 3 > 0 \Rightarrow K > 3/2$$

For sustained oscillations i.e. marginally stable system (Complete row zero)

$$\frac{K+3a+3}{3} = 0$$

But Both  $K, a > 0$

$$\Rightarrow \frac{K+3a+3}{3} \neq 0$$

Hence marginal stable not possible.



## Test Series: Control System with Solution

Num. of asymptotes =  $n - m = 1$

Angle of Asymptote  $\varphi_A = 180^\circ$

Break away point

$$K = -\frac{s^2 + 2s + 2}{s + 2}$$

$$\frac{dK}{ds} = -\frac{(s+2)(2s+2) - (s^2 + 2s + 2) \cdot 1}{(s+2)^2} = 0$$

$$\Rightarrow s^2 + 4s + 2 = 0$$

$$\Rightarrow s = -0.6, -3.4$$

$s = -0.6$  does not lie on root locus

For  $s = -3.4$  break away point

$$K = -\frac{(-3.4)^2 + 2 \times (-3.4) + 2}{-3.4 + 2}$$

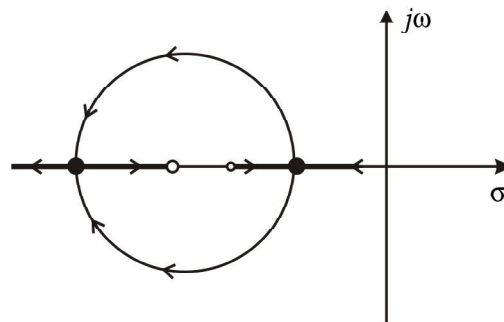
$$= 4.83$$

At  $K = 4.83$  both roots are on -ve real axis, i.e. coincident means the system is critically damped

if  $0 < K < 4.83$  Complex conjugate roots Hence the response is under damped

If  $K > 4.83$  Both roots on -ve real axis and distinct hence the response is over damped

**Q.16** The root locus of the control system  $1 + \frac{K(s+4)}{s(s+3)} = 0$  is given as



The centre & radius of the circle in root locus is given as

(a)  $(-5, 0)$  & 3

(b)  $(-6.5, 0)$  & 4.5

(c)  $(-4, 0)$  & 2

(d)  $(-8, 0)$  & 6

**Ans. (c)**

**Exp.**  $\frac{K(s+4)}{s(s+3)} = 1$

$$s = \sigma + j\omega$$

$$\frac{K[(4+\sigma)+j\omega]}{(\sigma+j\omega)[(3+\sigma)+j\omega]} = -1 = 1 \angle 180^\circ$$

$$\Rightarrow \frac{(\sigma+4)+j\omega}{(\sigma+j\omega)[(\sigma+3)+j\omega]} = 180^\circ$$

$$\Rightarrow \angle(\sigma+4)+j\omega - [\angle(\sigma+j\omega) + \angle(\sigma+3)+j\omega] = 180^\circ$$

$$\Rightarrow \tan^{-1}\left(\frac{\omega}{\sigma+4}\right) - \tan^{-1}\left(\frac{\omega}{\sigma+3}\right) = 180^\circ + \tan^{-1}\frac{\omega}{\sigma}$$

$$\Rightarrow \tan^{-1}\left[\frac{\frac{\omega}{\sigma+4} - \frac{\omega}{\sigma+3}}{1 + \frac{\omega^2}{(\sigma+3)(\sigma+4)}}\right] = 180^\circ + \tan^{-1}\frac{\omega}{\sigma}$$

$$\frac{\omega(\sigma+3) - \omega(\sigma+4)}{(\sigma+3)(\sigma+4) + \omega^2} = \frac{\omega}{\sigma}$$

$$\Rightarrow -\sigma = \sigma^2 + 7\sigma + 12 + \omega^2$$

$$\Rightarrow \sigma^2 + 8\sigma + \omega^2 + 12 = 0$$

$$\Rightarrow (\sigma+4)^2 + \omega^2 = 4$$

Centre of circle  $(-4, 0)$  radius = 2

**Q.17** For a unity feedback system the OLTF  $G(s) = \frac{1}{s(1+s)^2}$ .

The gain margin of system is

- (a) 0.5 (b) 2  
(c) 1 (d) 0

**Ans. (b)**

**Exp.** Frequency response  $s + j\omega$

$$G(j\omega) = \frac{1}{j\omega(1+j\omega)^2}$$

Gain

$$M = |G(j\omega)| = \frac{1}{\omega(1+\omega^2)}$$

$$\text{Phase } \varphi = \angle G(j\omega) = -90^\circ - 2 \tan^{-1} \omega$$

At phase cross over freq.  $\omega_p$   $\varphi = -180^\circ$

$$-180^\circ = -90^\circ - 2 \tan^{-1} \omega_p$$

$$\Rightarrow 2 \tan^{-1} \omega_p = 90^\circ$$

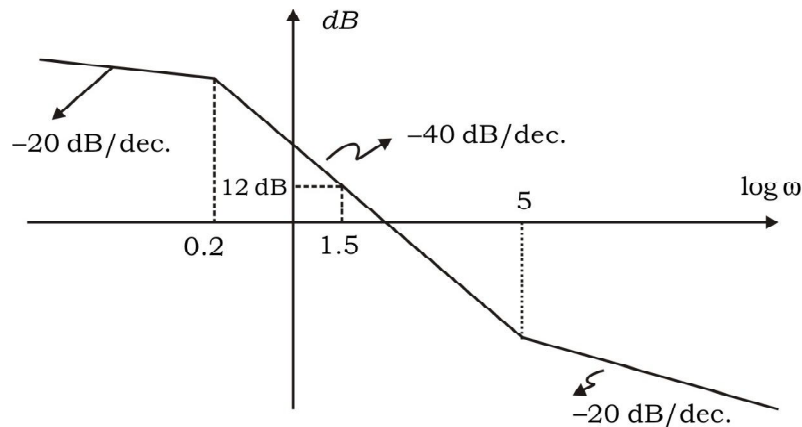
$$\Rightarrow \omega_p = 1$$

Gain at  $\omega_p = 1$

$$M = \frac{1}{1 \cdot (1+1^2)} = \frac{1}{2}$$

Gain margin  $GM = \frac{1}{\text{Gain at } \omega_p} = \frac{1}{1/2} = 2$

**Q.18** The gain plot for a minimum phase function is represented as



The T.F. of the system is (approximately)

(a)  $\frac{3(s+5)}{s^2(5s+1)}$

(b)  $\frac{15(5s+1)}{s(s+5)}$

(c)  $\frac{9(s+5)}{s(5s+1)}$

(d)  $\frac{12(s+5)}{s^2(5s+1)}$

**Ans. (c)**

**Exp.**

The corner frequencies are  $0.2 \text{ rad./sec}$  &  $5 \text{ rad/sec}$ .

Lowest corner freq.  $\omega_{cl} = 0.2 \text{ rad/sec}$

The initial slope, i.e. slope at frequencies less than the lowest corner freq.

$$= -20 \text{ dB / dec}$$

so its is Type - 1 system

so a term of  $\frac{K}{s}$  exists

$$G(s) = \frac{K (1+s/5)}{s (1+s/0.2)}$$

$$G(j\omega) = K \frac{\left(1 + j \frac{\omega}{5}\right)}{j\omega \left[1 + j \frac{\omega}{0.2}\right]}$$

At  $\omega < 5$   
Gain in dB

$$dB = 20 \log |G(j\omega)| = 20 \log K - 20 \log \omega - 20 \log \left[ 1 + \left( \frac{\omega}{0.2} \right)^2 \right]^{\frac{1}{2}}$$

$$dB = 20 \log K - 20 \log \omega - 20 \log (5\omega)$$

$$= 20 \log K - 40 \log \omega - 20 \log 5$$

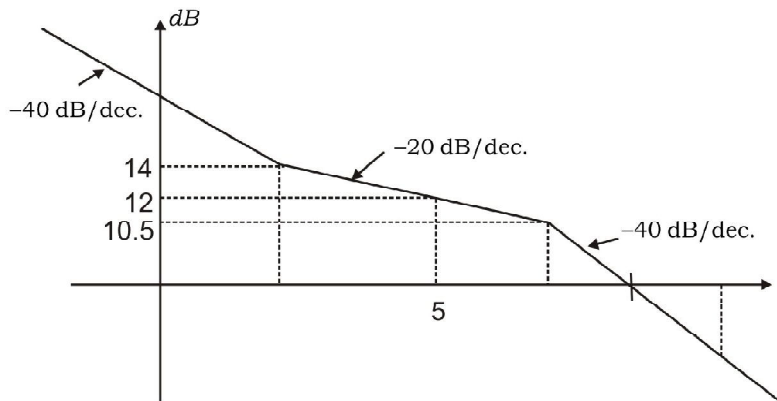
At  $\omega = 1.5$ ,  $dB = 12$

$$12 = 20 \log K - 40 \log 1.5 - 20 \log 5$$

$$K = 44.8 \approx 45$$

$$G(S) = \frac{45(1+s/5)}{s(1+5s)} = \frac{9(s+5)}{s(5s+1)}$$

Q.19



The T.F. of minimum phase function

(a)  $\frac{100(s+2)}{s^2(s+6)}$

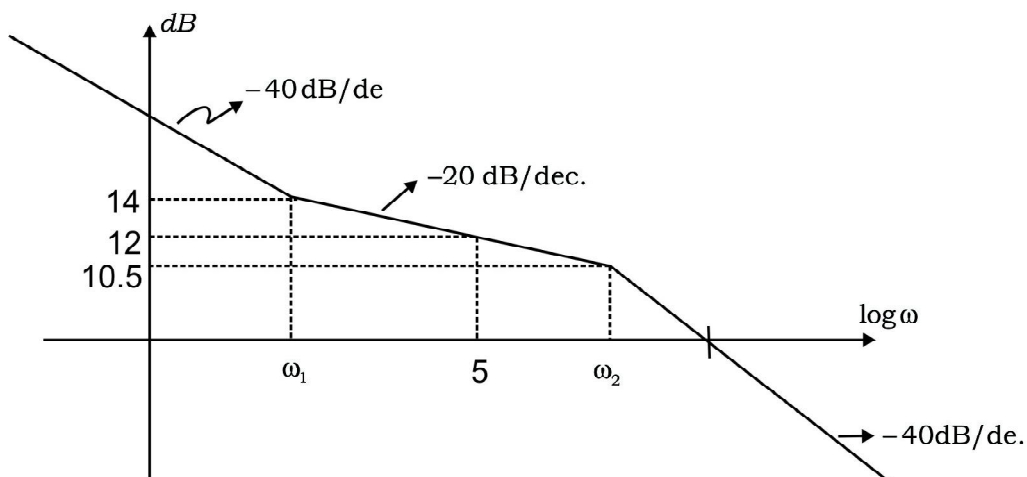
(b)  $\frac{120(s+2)}{s^2(s+3)}$

(c)  $\frac{120(s+4)}{s^2(s+5)}$

(d)  $\frac{120(s+4)}{s^2(s+6)}$

Ans. (d)

Exp.



The corner freq. are  $\omega_1$  &  $\omega_2$

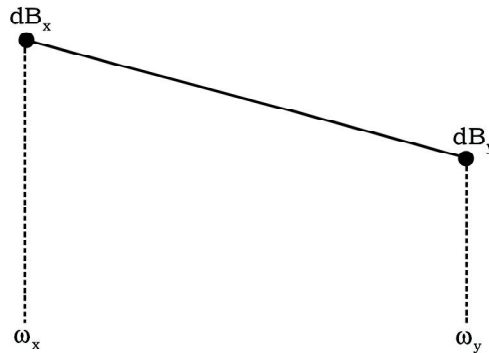
Initial slope =  $-40 \text{ dB / dec}$

so it is Type-2 system

mean a term of  $K / s^2$

$$G(S) = \frac{K (1 + s / \omega_1)}{s^2 (1 + s / \omega_2)}$$

If slope of line M



$$dB_x - dB_y = M \log_{10} \frac{\omega_x}{\omega_y}$$

For  $M = -20 \text{ dB / dec.}$

$$\omega_x = \omega_1, \quad dB_x = 14$$

$$\omega_y = 5, \quad dB_y = 12$$

$$14 - 12 = -20 \log_{10} \frac{\omega_1}{5}$$

$$\Rightarrow \omega_1 \approx 4 \text{ rad/sec.}$$

$$G(s) = \frac{K(1 + s/4)}{s^2(1 + s/\omega_2)}$$

At  $\omega \leq 4$

$$G(s) \approx \frac{K}{s^2} = \frac{K}{(j\omega)^2}$$

$$dB = 20 \log K - 40 \log \omega$$

At  $\omega = \omega_1 = 4, \quad dB = 14$

$$14 = 20 \log K - 40 \log 4$$

$$\Rightarrow K \approx 80$$

$$G(s) = \frac{80(1 + s/4)}{s^2(1 + s/\omega_2)}$$

## Test Series: Control System with Solution

At  $\omega_x = 5, dB_x = 12$

$\omega_y = \omega_2 \quad dB_y = 10.5$

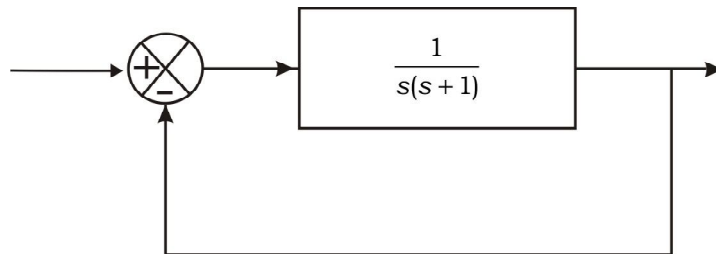
$$12 - 10.5 = -20 \log \frac{5}{\omega_2}$$

$\Rightarrow \omega_2 \approx 6$

$$G(S) = \frac{80(1+s/4)}{s^2(1+s/6)}$$

$$= \frac{120(s+4)}{s^2(s+6)}$$

**Q.20** For the control system shown



The sensitivity of CLTF wrt OLTF at  $\omega=1$  rad/sec is

(a)  $\sqrt{2}$

(b) 0.2

(c)  $\frac{1}{\sqrt{2}}$

(d) 2

**Ans. (a)**

**Exp.**  $OLTF \quad G = \frac{1}{s(s+1)}$

$$CLTF \quad T = \frac{G}{1+G}$$

$$S_G^T = \frac{\partial T / T}{\partial G / G} = \frac{G}{T} \frac{\partial T}{\partial G}$$

$$= \frac{1}{1+G} = \frac{1}{1 + \frac{1}{s(s+1)}}$$

$$= \frac{s^2 + s}{s^2 + s + 1}$$

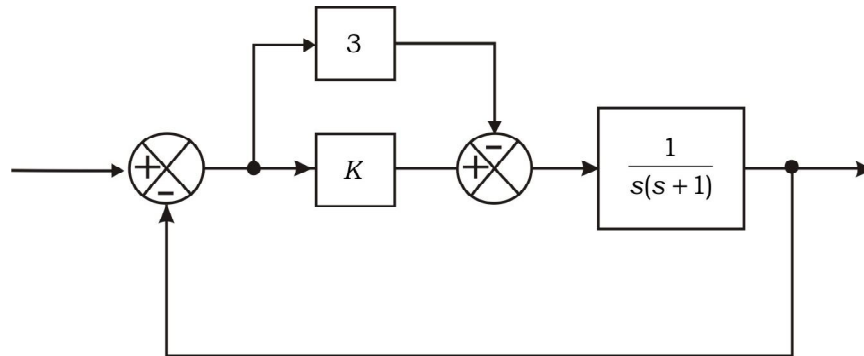
$$s = j\omega$$

$$S_G^T(j\omega) = \frac{j\omega - \omega^2}{1 - \omega^2 + j\omega}$$

At  $\omega = 1$

$$S_G^T(j1) = \frac{-1+j}{j} = 1+j = \sqrt{2} \angle 45^\circ$$

Q.21



The maximum possible sensitivity of CLTF wrt OLTF is at  $\omega = 1$

- (a) 2  
(b)  $\sqrt{\frac{2}{17}}$   
(c) 1  
(d)  $\sqrt{2}$

Ans. (d)

Exp.

$$OLTF \quad G(s) = \frac{K-3}{s(s+1)}$$

Ch.equation  $1+G(s) = 0$

$$s^2 + s + K - 3 = 0$$

For stable operation  $K > 3$

$$S_G^T = \frac{1}{1+G} = \frac{1}{1 + \frac{K-3}{s(s+1)}}$$

$$= \frac{s^2 + s}{s^2 + s + K - 3}$$

$$s = j1$$

$$S_G^T(j1) = \frac{-1+j}{-1+j+K-3} = \frac{-1+j}{(K-4)+j}$$

$$|S_G^T| = \sqrt{\frac{2}{(K-4)^2 + 1}}$$

For  $|S_G^T|_{\max} \quad K-4 = 0$   
 $\Rightarrow K = 4$

$$|S_G^T| = \sqrt{2}$$

**Common Data for Q.22 & 23**

The state space representing matrix are given

$$A = \begin{bmatrix} 0 & 1 \\ -1 & 0 \end{bmatrix}, B = \begin{bmatrix} 0 \\ 2 \end{bmatrix}, C = [1 \ 0]$$

In initial state  $x(0) = \begin{bmatrix} 0 \\ 0 \end{bmatrix}$

**Q.22** The state transition matrix

(a)  $\begin{bmatrix} \sin t & \cos t \\ -\cos t & \sin t \end{bmatrix}$

(b)  $\begin{bmatrix} \sin t & \cos t \\ \cos t & -\sin t \end{bmatrix}$

(c)  $\begin{bmatrix} \cos t & \sin t \\ -\sin t & \cos t \end{bmatrix}$

(d)  $\begin{bmatrix} \cos t & -\sin t \\ -\sin t & \cos t \end{bmatrix}$

**Ans. (c)**

**Q.23** The response for unit step input is

(a)  $\sin t + \cos t - 2$

(b)  $2(1 - \cos t)$

(c)  $2(1 + \cos t)$

(d)  $\sin t + t$

**Ans. (b)**

**Exp.**

$$\begin{aligned} [sI - A] &= s \begin{bmatrix} 1 & 0 \\ 0 & 1 \end{bmatrix} - \begin{bmatrix} 0 & 1 \\ -1 & 0 \end{bmatrix} \\ &= \begin{bmatrix} s & -1 \\ 1 & s \end{bmatrix} \end{aligned}$$

$$\phi(s) = [sI - A]^{-1} = \begin{bmatrix} s & -1 \\ 1 & s \end{bmatrix}^{-1}$$

$$= \frac{1}{s \cdot s - (1)(-1)} \begin{bmatrix} s & 1 \\ -1 & s \end{bmatrix}$$

$$= \begin{bmatrix} \frac{s}{s^2 + 1} & \frac{1}{s^2 + 1} \\ -\frac{1}{s^2 + 1} & \frac{s}{s^2 + 1} \end{bmatrix}$$

S.T.M

$$\phi(t) = \mathcal{L}^{-1}[\phi(s)]$$

$$= \begin{bmatrix} \cos t & \sin t \\ -\sin t & \cos t \end{bmatrix}$$

$$X(s) = \phi(s)x(0) + \phi(s)BU(s)$$

$$= \frac{1}{s^2+1} \begin{bmatrix} s & 1 \\ -1 & s \end{bmatrix} \begin{bmatrix} 0 \\ 2 \end{bmatrix} \frac{1}{s}$$

$$= \frac{1}{s(s^2+1)} \begin{bmatrix} 2 \\ 2s \end{bmatrix} = \frac{2}{s(s^2+1)} \begin{bmatrix} 1 \\ s \end{bmatrix}$$

output  $Y(s) = C X(s)$

$$= [1 \quad 0] \begin{bmatrix} \frac{2}{s(s^2+1)} \\ \frac{2}{s^2+1} \end{bmatrix}$$

$$= \frac{2}{s(s^2+1)} = \frac{2}{s} - \frac{2s}{s^2+1}$$

$$Y(t) = 2 - 2\cos t$$

**Q.24** For a system with the transfer function

$G(s) = \frac{2(s-3)}{s^3+2s^2-3s+1}$ , the matrix A in the state space from  $\dot{x} = Ax + Bu$  is equal to

(a)  $\begin{bmatrix} 0 & 1 & 0 \\ 0 & 0 & 1 \\ -1 & 3 & -2 \end{bmatrix}$

(b)  $\begin{bmatrix} 0 & 1 & 0 \\ 0 & 0 & 1 \\ 2 & -3 & 6 \end{bmatrix}$

(c)  $\begin{bmatrix} 0 & 1 & 0 \\ 0 & 0 & 1 \\ 1 & -3 & 2 \end{bmatrix}$

(d)  $\begin{bmatrix} 0 & 1 & 0 \\ 0 & 0 & 1 \\ -2 & 6 & 3 \end{bmatrix}$

**Ans. (a)**

**Exp.**

$$G(s) = \frac{Y(s)}{U(s)} = \frac{2s-6}{s^3+2s^2-3s+1}$$

$$Y(s) = -6 \cdot \frac{1}{s^3+2s^2-3s+1} U(s) + 2s \cdot \frac{1}{s^3+2s^2-3s+1} U(s)$$

Let  $Y_1(s) = \frac{1}{s^3+2s^2-3s+1} U(s)$

$$s^3 Y_1 + 2s^2 Y_1 - 3s Y_1 + Y_1 = U(s)$$

$$\Rightarrow \overset{\dots}{y} + 2\overset{\dots}{y}_1 - 3\overset{\cdot}{y}_1 + y_1 = u$$

Let  $y_1 = x_1, \dot{y}_1 = x_2, \ddot{y}_1 = x_3$

$$\Rightarrow \dot{x}_1 = x_2$$

$$\dot{x}_2 = x_3$$

$$\dot{x}_3 + 2x_3 - 3x_2 + x_1 = u$$

$$\Rightarrow \dot{x}_3 = -x_1 + 3x_2 - 2x_3 + 4$$

$$\begin{bmatrix} \dot{x}_1 \\ \dot{x}_2 \\ \dot{x}_3 \end{bmatrix} = \begin{bmatrix} 0 & 1 & 0 \\ 0 & 0 & 1 \\ -1 & 3 & -2 \end{bmatrix} \begin{bmatrix} x_1 \\ x_2 \\ x_3 \end{bmatrix} + \begin{bmatrix} 0 \\ 0 \\ 1 \end{bmatrix} u$$

$$Y(s) = -6Y_1 + 2sY_1$$

$$y = -6y_1 + 2\dot{y}_1$$

$$y = -6x_1 + 2x_2$$

**Q.25** A unity feedback control system has a forward loop T.F. as  $\frac{e^{-s/2}}{1+s}$ . The phase cross over frequency is given by

(a)  $\omega = \cot\left(\frac{\omega}{2}\right)$

(b)  $2\omega = \tan \omega$

(c)  $2\omega + \cot \omega = 0$

(d)  $\omega + \tan\left(\frac{\omega}{2}\right) = 0$

**Ans. (d)**

**Exp.**  $G(s) = \frac{e^{-s/2}}{1+s}$

$$G(j\omega) = \frac{e^{-j\omega/2}}{1+j\omega}$$

phase  $\phi = \angle G(j\omega) = -\frac{\omega}{2} - \tan^{-1} \omega = -180^\circ$

$$\tan^{-1}(\omega) = 180^\circ - \frac{\omega}{2}$$

$$\omega = \tan(180^\circ - \omega/2)$$

$$\Rightarrow \omega = -\tan(\omega/2)$$

$$\Rightarrow \tan(\omega/2) + \omega = 0$$



(a)  $a > 2$

(b)  $a < 2$

(c)  $a < 0$

(d)  $a > 0$

**Ans. (c)**

**Exp.**

⇒ Ch. equation

$$|sI - A| = 0$$

$$sI - A = s \begin{bmatrix} 1 & 0 \\ 0 & 1 \end{bmatrix} - \begin{bmatrix} 0 & 1 \\ -2 & a \end{bmatrix} = \begin{bmatrix} s & -1 \\ 2 & s-a \end{bmatrix}$$

$$|sI - A| = \begin{vmatrix} s & -1 \\ 2 & s-a \end{vmatrix} = s(s-a) + 2 = 0$$

Ch. equation

$$s^2 - as + 2 = 0$$

$$s^2 \quad \left| \begin{array}{cc} 1 & 2 \end{array} \right.$$

$$s^1 \quad \left| \begin{array}{cc} -a & 0 \end{array} \right.$$

$$s^0 \quad \left| \begin{array}{cc} 2 & 0 \end{array} \right.$$

For stability  $-a > 0$

⇒  $a < 0$

**Q.28** For a lead network having T.F. as  $\frac{1+0.1s}{1+0.04s}$ . The maximum phase shift is

(a)  $38.6^\circ$

(b)  $25.4^\circ$

(c)  $22.8^\circ$

(d)  $30^\circ$

**Ans. (b)**

**Exp.**

$$G_c(s) = \frac{1+0.1s}{1+0.04s}$$

Compare with  $G_c(s) = \frac{1+s\tau}{1+s\alpha\tau}$

$$\tau = 0.1 \quad \alpha\tau = 0.04$$

⇒  $\alpha = 0.4$

Maximum phase shift

$$\tan \phi_m = \frac{1-\alpha}{2\sqrt{\alpha}}$$

$$= \frac{1-0.4}{2\sqrt{0.4}}$$

⇒  $\phi_m = 25.4^\circ$

**Q.29** For *ufb* OLTF

$$G(s) = \frac{2e^{-s}}{2s+1}$$

The phase margin is

(a)  $70.15^\circ$

(b)  $-109.85^\circ$

(c)  $109.85^\circ$

(d)  $250.85^\circ$

**Ans. (a)**

**Exp.**

$$G(j\omega) = \frac{2e^{-j\omega}}{1+j2\omega}$$

Gain

$$M = |G(j\omega)| = \frac{2}{\sqrt{1+(2\omega)^2}}$$

At gain cross-over freq.  $\omega_g$

$$M = 1$$

$$\frac{2}{\sqrt{1+(2\omega_g)^2}} = 1$$

$$\Rightarrow 1 + (2\omega_g)^2 = 4$$

$$\Rightarrow \omega_g = 0.87 \text{ rad/sec.}$$

$$\varphi = \angle G(j\omega) = -3\omega - \tan^{-1}(2\omega)$$

At  $\omega_g$

$$\varphi = -\omega_g - \tan^{-1}(2\omega_g)$$

$$= -0.87 - \tan^{-1}(1.73)$$

$$= -109.85^\circ$$

$$GM = 180^\circ + \varphi = 70.15^\circ$$

**Q.30** For a second order system the damping ratio is 0.7 the PM of system (approximately) is

(a)  $30^\circ$

(b)  $40^\circ$

(c)  $70^\circ$

(d)  $56^\circ$

**Ans. (c)**

**Exp.**

PM (approx)

$$PM \approx 100\rho \text{ (degrees)}$$

$$PM \approx 100 \times 0.7$$

$$\approx 70^\circ$$